

Amendments to the Claims:

This listing of claims will replace all prior versions, and listings, of the claims in the application:

Listing of Claims:

1. (Currently Amended) ~~Improved~~ An assembly, comprising: ~~(110) consisting of an~~ internal casing ~~(112)~~ and a support device ~~(114)~~ for nozzles in a gas turbine stage, said nozzles being grouped together in sectors and each of said sectors being connected externally to an external casing of said gas turbine by means of said support device ~~(114)~~, said support device ~~(114)~~ being kept in position by said internal casing ~~(112)~~, there also being formed first cooling holes ~~(122)~~ on said internal casing ~~(112)~~ and second cooling holes ~~(124)~~ on said support device ~~(114)~~, ~~characterized in that~~ wherein said first cooling holes ~~(122)~~ of said internal casing ~~(112)~~ have an extension substantially parallel to the axis of said gas turbine, wherein said first holes are arranged circumferentially and are forty-two in number.
2. (Currently Amended) ~~Improved~~ An assembly ~~(110)~~ according to Claim 1, ~~characterized in that~~ wherein said support device ~~(114)~~ has ~~internally a~~ an internal cooling recess ~~(126)~~.
3. (Currently Amended) ~~Improved~~ An assembly ~~(110)~~ according to Claim 1, ~~characterized in that~~ wherein cooling inserts ~~(118)~~ are provided in said support ~~devices~~ device ~~(114)~~.
4. (Currently Amended) ~~Improved~~ An assembly ~~(110)~~ according to Claim 3, ~~characterized in that~~ wherein said cooling inserts ~~(118)~~ are brazed along an external diameter of said support ~~devices~~ device ~~(114)~~.
5. (Currently Amended) ~~Improved~~ An assembly ~~(110)~~ according to Claim 1, ~~characterized in that~~ wherein an anti-rotational pin ~~(120)~~ is provided, being located substantially at the front of said support device ~~(114)~~.

6. (Currently Amended) ~~Improved~~ An assembly-(110) according to Claim 1, ~~characterized in that~~ further comprising a contact surface-(128) supporting an axial thrust exists between said internal casing-(112) and said support device-(114).

7. (Currently Amended) ~~Improved~~ An assembly-(110) according to Claim 1, ~~characterized in that~~ wherein said support devices-(114) are grouped together in sectors.

8. (Currently Amended) ~~Improved~~ An assembly-(110) according to Claim 1, ~~characterized in that~~ wherein said support devices-(114) are kept in position by said internal casing (112) by means of grooves and pins and interlocking joints (116)-with said nozzles.

9. (Currently Amended) ~~Improved~~ An assembly-(110) according to Claim 1, ~~characterized in that~~ wherein said second cooling holes (124) are arranged at the rear of said support device-(114).

10. (Canceled)

11. (Currently Amended) ~~Improved~~ An assembly-(110) according to Claim 1, ~~characterized in that~~ wherein said first holes-(122) each have an approximate diameter of 1.8 mm.

12. (Currently Amended) ~~Improved~~ An assembly-(110) according to Claim 1, ~~characterized in that~~ wherein said stage is the first high-pressure stage of a gas turbine.